An aerial photograph of the UCSB coastline, showing the ocean, a sandy beach, and the university campus buildings. A large, semi-transparent white circle is overlaid on the center of the image. Inside this circle, the UCSB logo is displayed, followed by the word 'HYPER' in a large, thin, sans-serif font. Below 'HYPER' is a stylized graphic of a lens or a camera aperture, and the letter 'P' is positioned to the right of the graphic.

UCSB
HYPER
P

Final Presentation
Spring 2017

SENSORS & CONTROLS



CEs: Yang Ren, Jesus Diera, Asitha Kaduwela, Tristan Seroff

EEs: Brian Canty, Ricardo Castro, Kevin Kha

Team

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Structure

Stabilization

Levitation

Braking

Electronics

Power

Thermal

Logistics

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Actuators

Navigation

Power

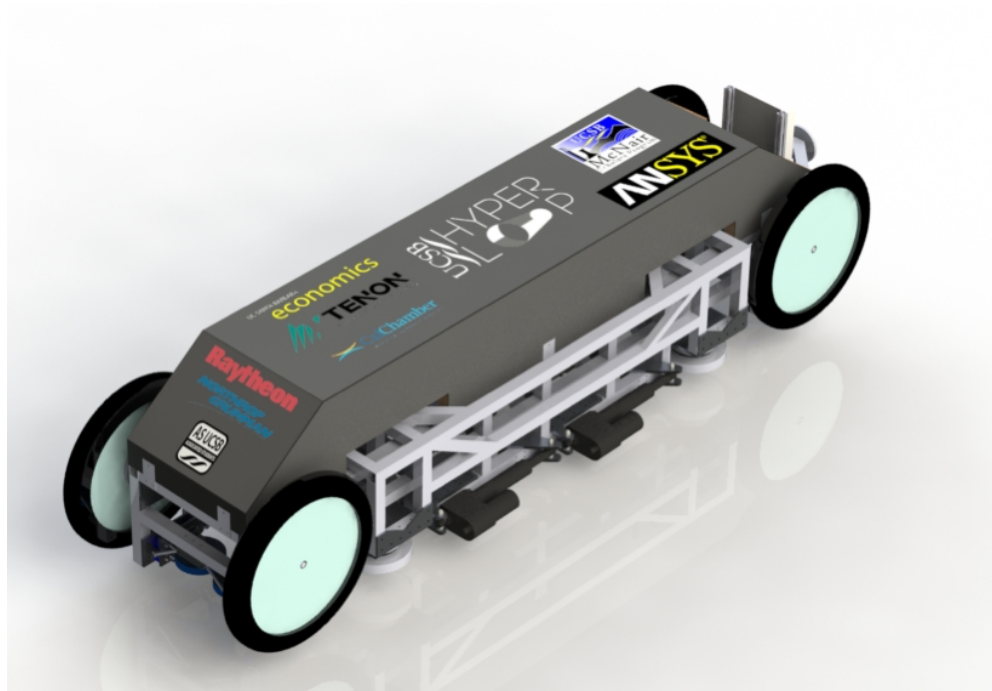
Control Systems

Pod Run

Testing

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- Overview
- Hardware
- Maglev
- Actuators
- Navigation
- Power
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- Conclusion



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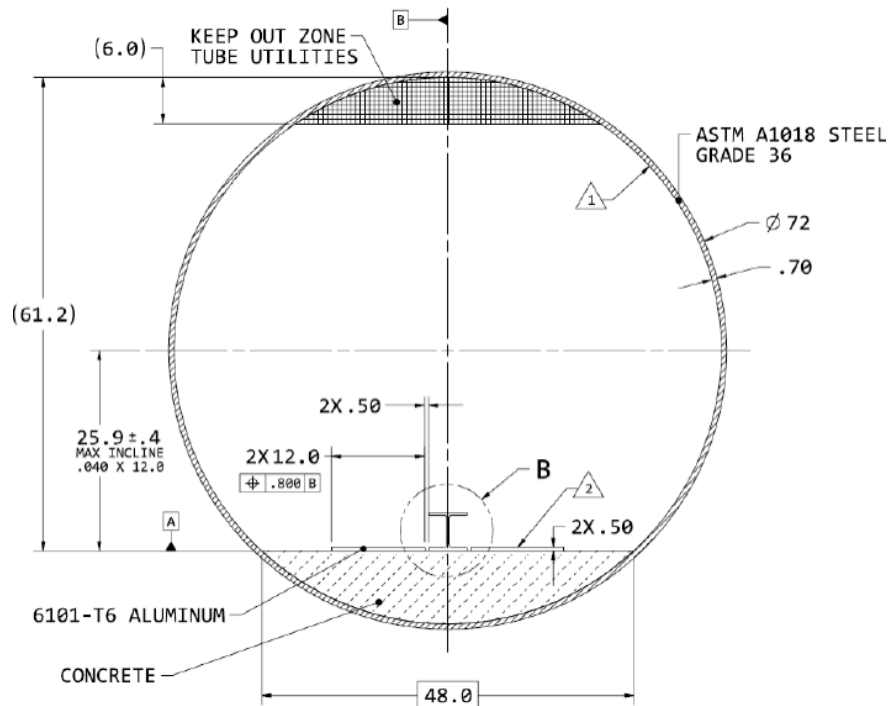
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SPACEX COMPETITION

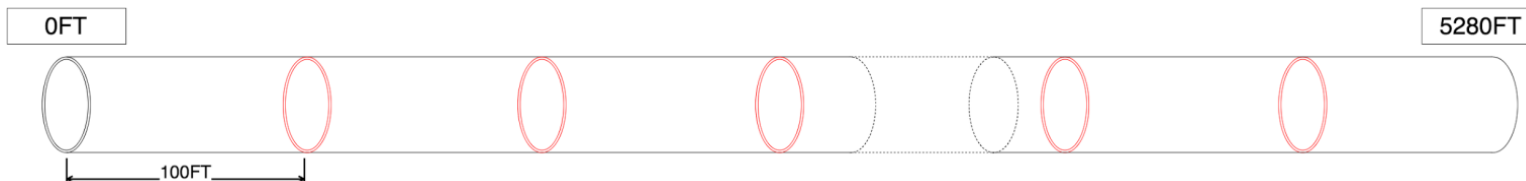
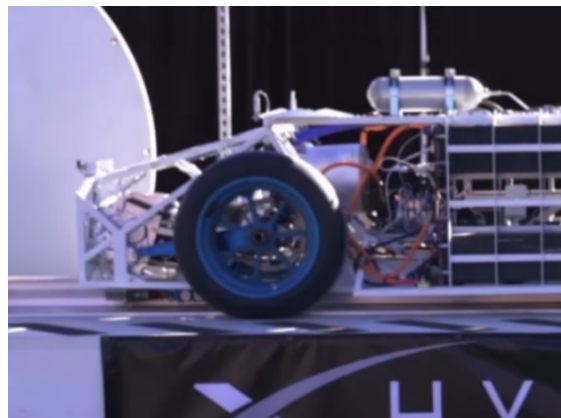
- SpaceX has constructed a mile long test track to develop technology through open competition
- Task: Design a Hyperloop Pod
- Main criteria of success: top pod speed
 - Our pod's max theoretical speed is 205mph



SPACEX COMPETITION

Reducing project complexity

- Photo-reflective tape strips every 100 ft for navigation
- Pod pusher for propulsion
- I-Beam for stabilization
- Network Access Panel for communications



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DESIGN SUMMARY

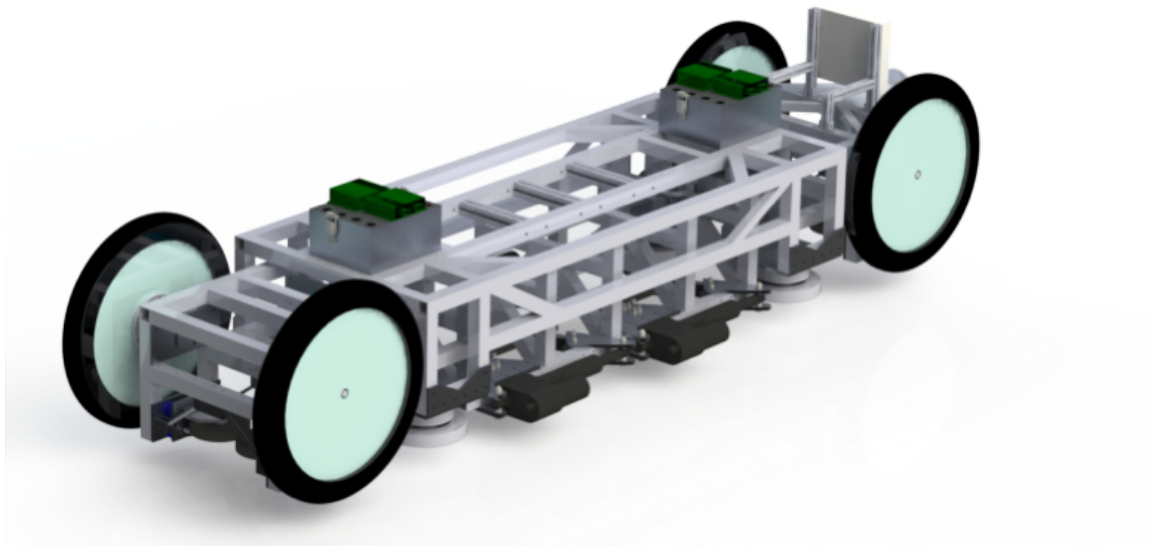


Figure: UCSB Hyperloop proposed design

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Hanson America (417) 731-0011

EAGLE

USA 32

GOODYEAR

STOP

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FRAME

- Made of Al-6061 Tubes
- Cart chassis (**green**)
 - Rides on drag wheels
 - Supports stabilization and braking
- Payload chassis (**blue**)
 - Levitates on maglev motors
 - Floats on linear bearings (**red**)
 - Supports the payload.

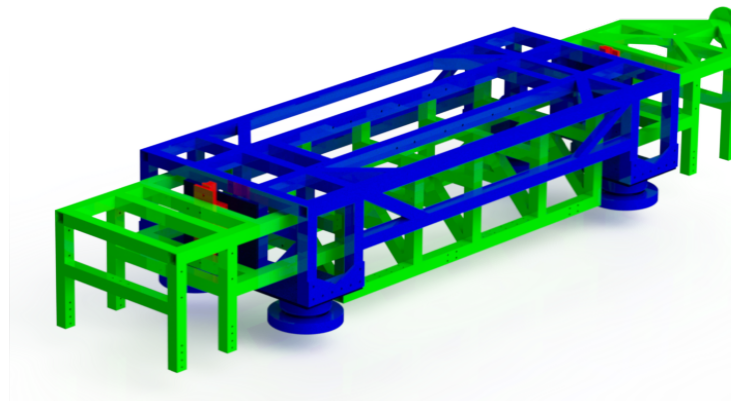


Figure: Pod Chassis

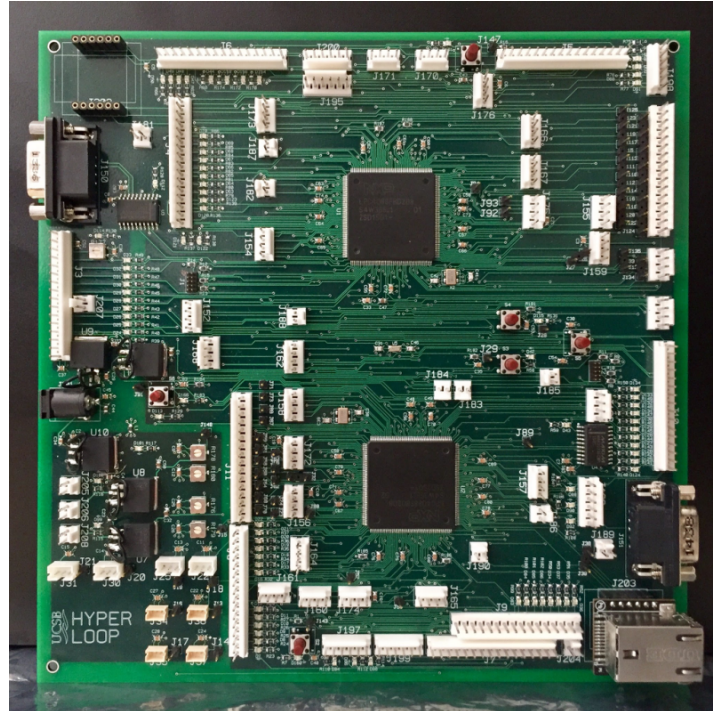
HARDWARE

Pod Control Board (PCB)

- Designed by the 2015-2016 team
- Two (2) LPC4088 chips
- Three (3) I2C ports
- Two (2) SPI ports
- Lots of GPIO + ADCs
- Ethernet module

PCB Code

- Pod control system
- Actuation & data collection
- Communication & telemetry w/ web app



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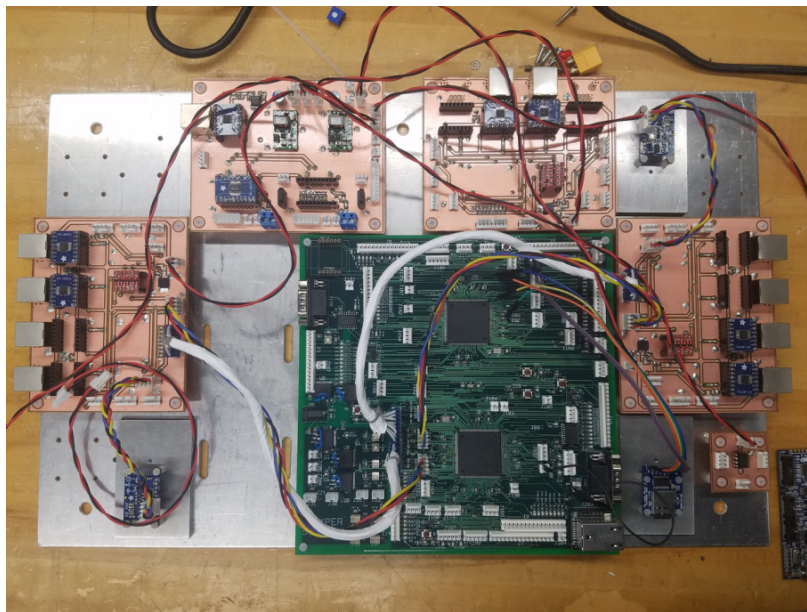
HARDWARE

Electronics Mounting

- PCB
- I2C hubs
- Power distribution board
- SD card breakout
- Accelerometers

Peripheral Boards (I2C)

- Types
 - Motor control
 - Actuator control
 - Battery management



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PAYLOAD LEVITATION

- Arx Pax HE3.0 Hover Engines lift payload
 - Brushless DC motor spins permanent magnet Halbach Array
- Hyperloop v2 team using four engines
 - Motor current does not exceed 50 A (continuous rating of motor controllers)
- Mounted to payload frame
 - Wheels serve as backup if levitation fails

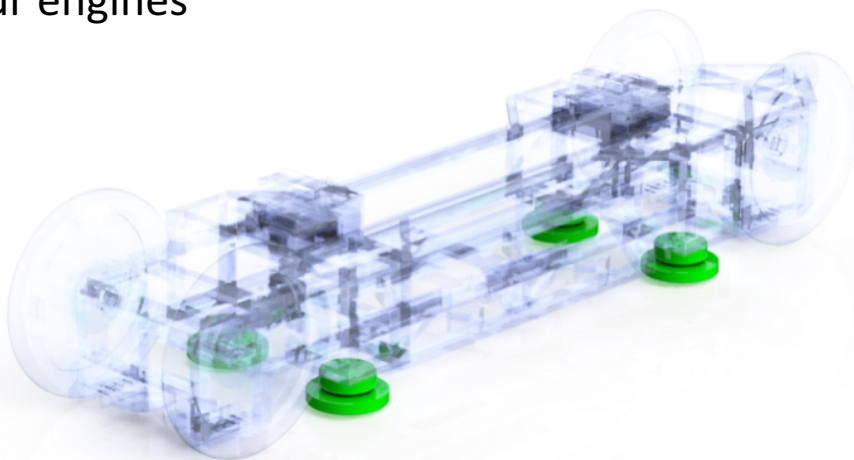


Figure: Position of hover engines on pod

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PAYLOAD LEVITATION

- Motor Control
 - Analog throttle signal provided to control motor speed
- Motor Sensors
 - Tachometers implemented to read RPM (0-2800 RPM)
 - Current and temperature sensors implemented as well
- I2C Motor Boards

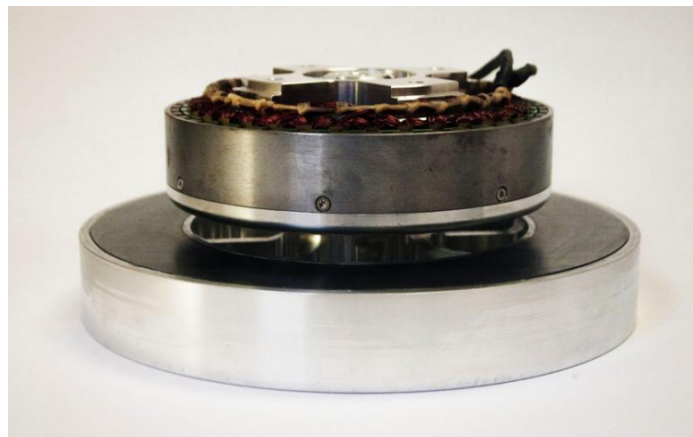


Figure: Arx Pax HE3.0 Hover Engine

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BRAKING

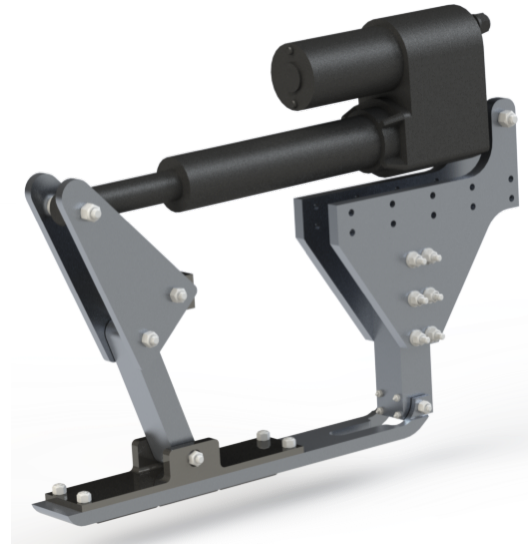
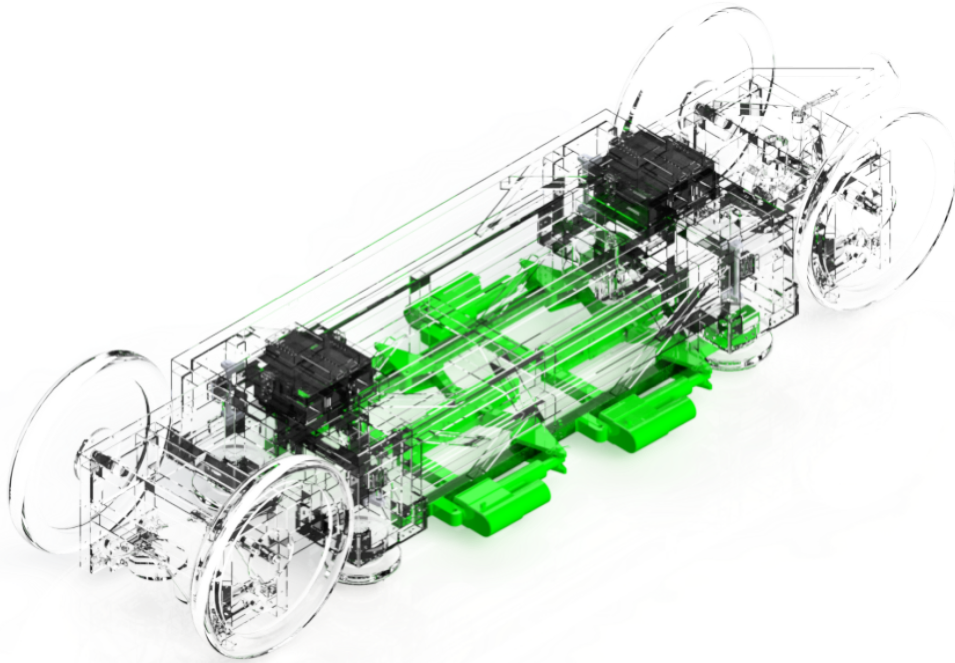


Figure: Brakes subsystem and placement on pod chassis

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BRAKING SPECIFICATIONS

- Friction Brakes - Two Pairs
 - Located at the front and rear
 - Equal distance from center of mass
 - Reduces moment on I-beam
- Brake Controls & Feedback
 - PWM speed control and direction signal
 - Position feedback through I2C ADC



Direction of motion

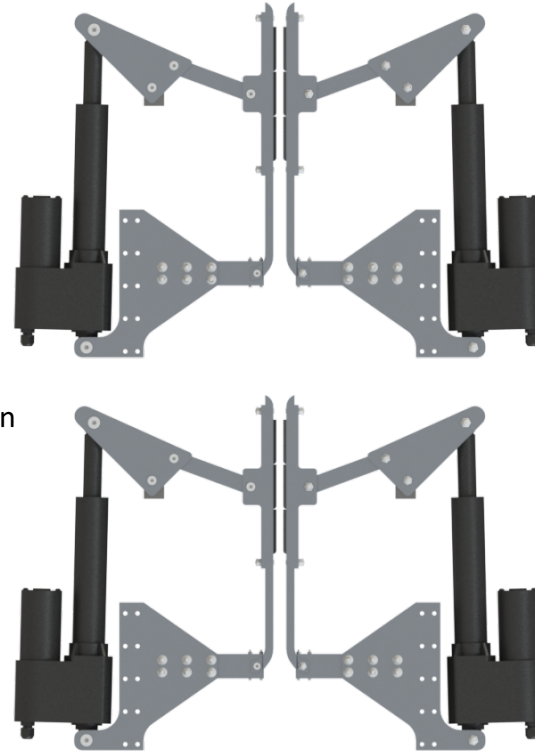


Figure: Top view of system

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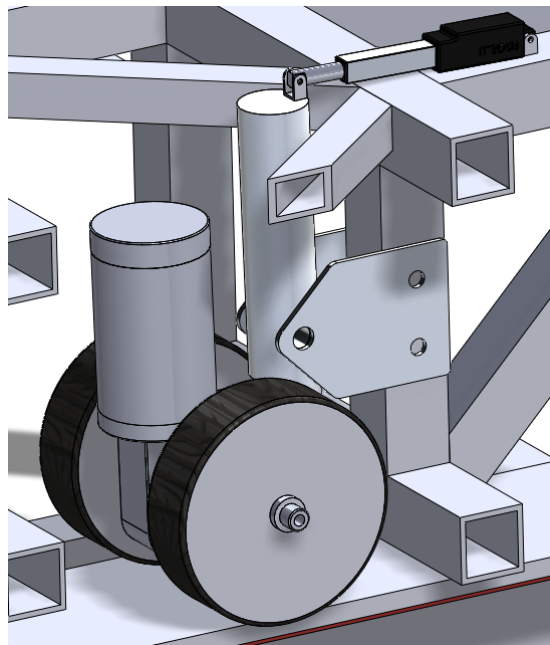
CONTACT SENSOR

- Pod Pusher Interface
- Optical Contact Sensor
 - Photo-reflective contact sensor indicates when pusher is mated with pod
 - 3-inch detection range

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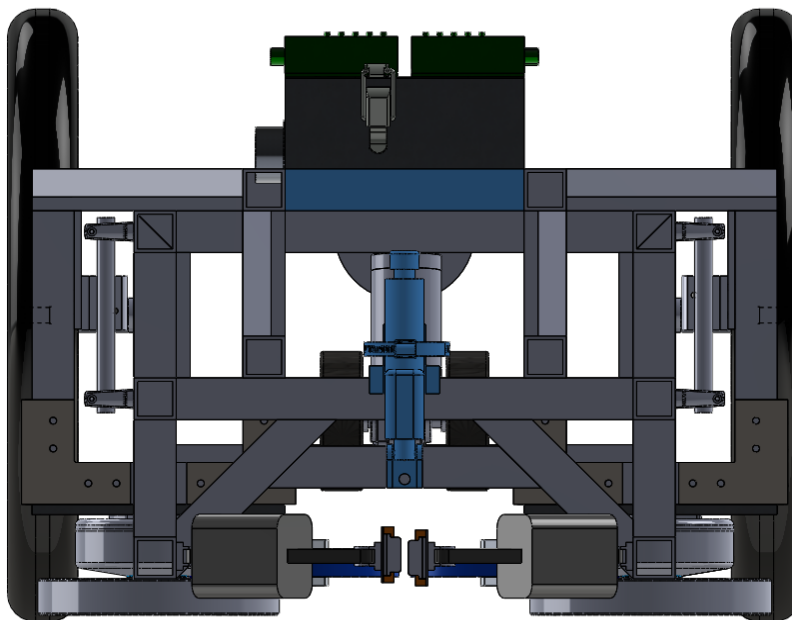
SERVICE PROPULSION

- Post-Braking
 - Service propulsion drive for low speed service travel powered internally
 - Used after successful braking when the pod is fully stopped

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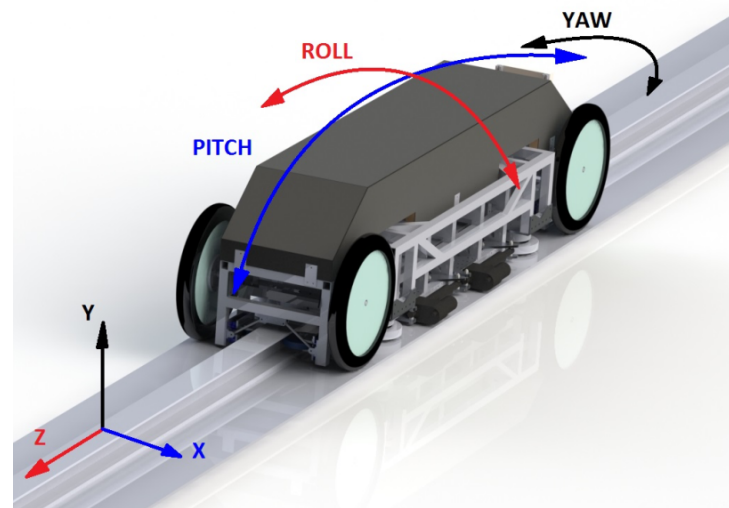
PAYLOAD ACTUATORS

- Vertical Service Lifter
 - Located opposite ends of pod
 - Raises payload to reduce magnetic drag for service propulsion
 - Uses linear actuators that push against the payload chassis

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NAVIGATION

Direction	Sensor System
X	I-Beam Short-Ranging
Y	Downward Short-Ranging
Z	Photoelectric Sensors Cart Wheel Tachometers Accelerometers
Roll	Downward Short-Ranging
Pitch	Downward Short-Ranging
Yaw	I-Beam Short-Ranging



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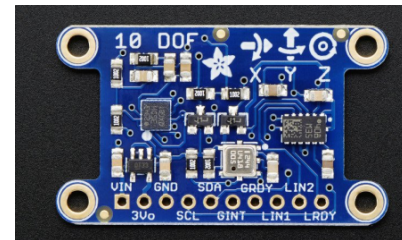
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Z-AXIS MOVEMENT

- Acceleration is measured directly with two accelerometers
- Distance is measured with photoelectric sensors that detect reflective strips placed 100 ft apart
- Velocity and distance are measured with tachometers placed on each of the four wheels



SHORT-RANGING

Downward facing (4)

- Measures the pod's vertical offset from the track
- Y-Position, Roll, and Pitch

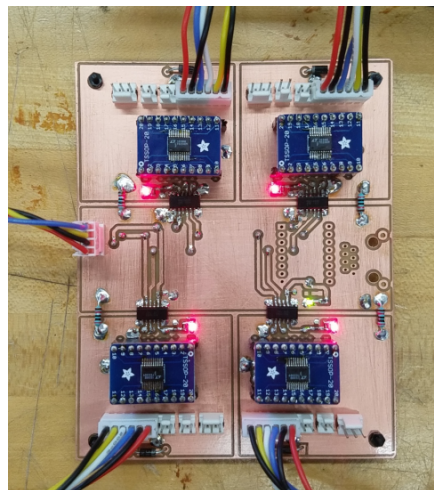
I-beam facing (2)

- Measures the pod's lateral offset from the I-Beam
- X-direction and Yaw

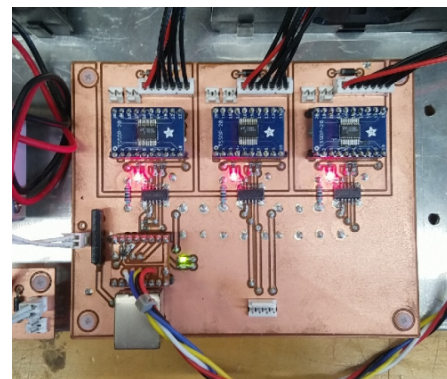
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BATTERY MANAGEMENT

- Custom designed and fabricated battery management boards for each subsystem
- Monitor battery voltage, cell voltages, current, battery temperature



5 Cell Battery BMS board



6 Cell Battery Maglev BMS board

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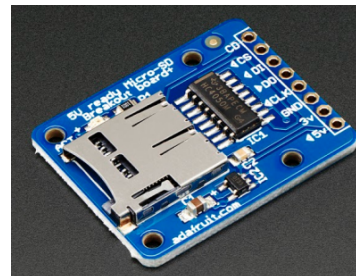
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DATA & TELEMETRY

- Telemetry
 - All sensor data automatically sent to web app and logged on SD card
 - Sensor data transmitted in real time at 10 Hz
- Control Signals
 - Sent from web app
 - Initialization
 - Start
 - Emergency braking



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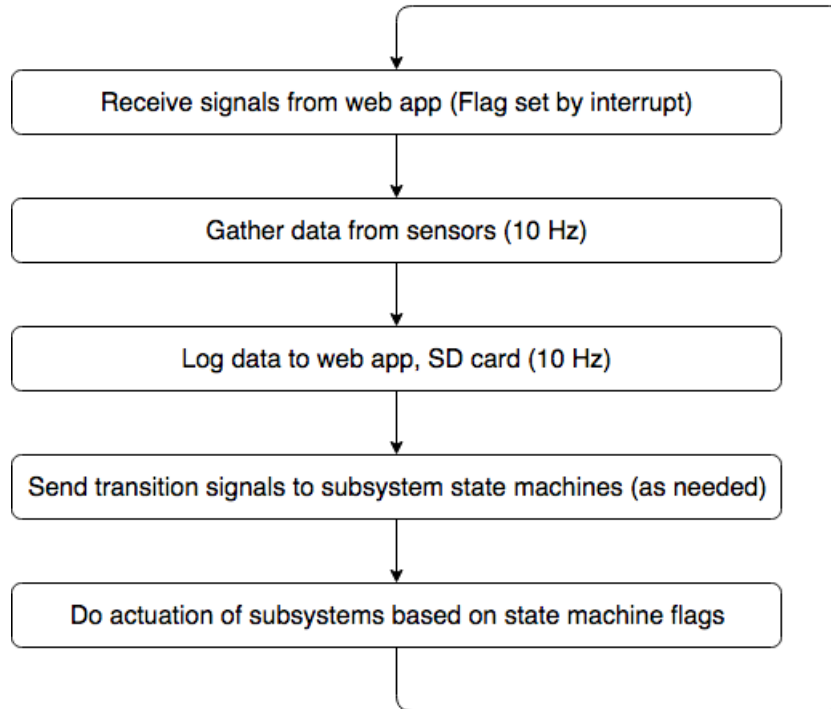
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SOFTWARE ARCHITECTURE



Special Routines:

- Braking feedback / control (high frequency TBD)
 - Only during braking phase (with feedback)
 - Braking force adjusted dynamically

- Photoelectric strip detected
 - Triggers interrupt
 - Navigation processes immediately

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STATE MACHINES

State machines guard actuation of subsystems

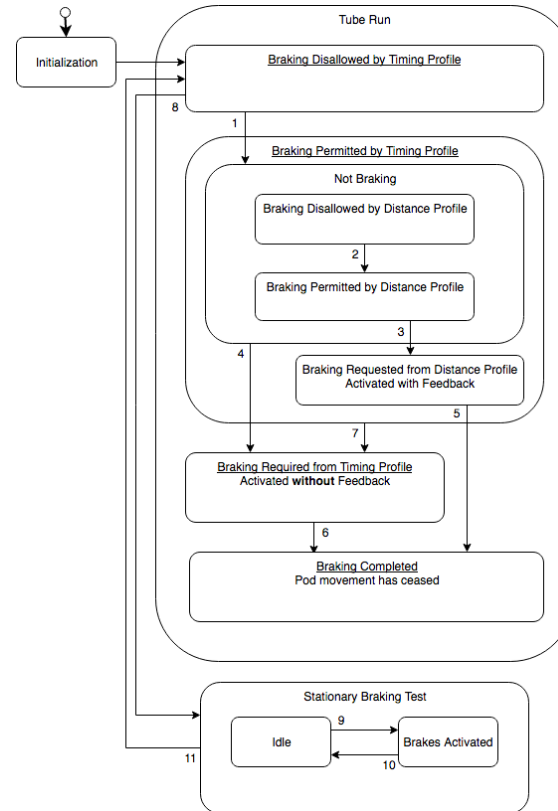
- States represent conditions of pod / system
- Signals issued based on sensor data
 - Ensure proper conditions are met
- Built on the QPnano framework

Subsystems with state machines

- Braking (see diagram)
- Magnetic levitation motors
- Payload actuators
- Service propulsion system

Subsystems without direct actuation don't need state machines

- Navigation
- Power



Braking Subsystem State Machine

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POD RUN @ SPACEX



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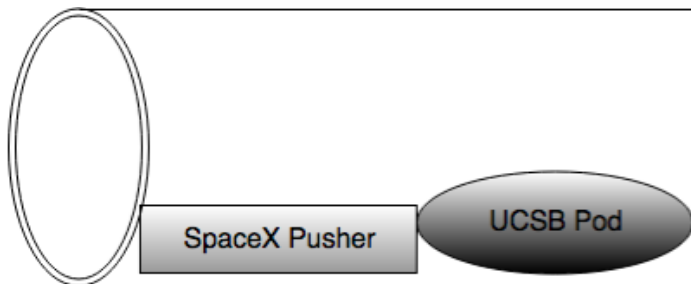
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PRE-RUN PHASE



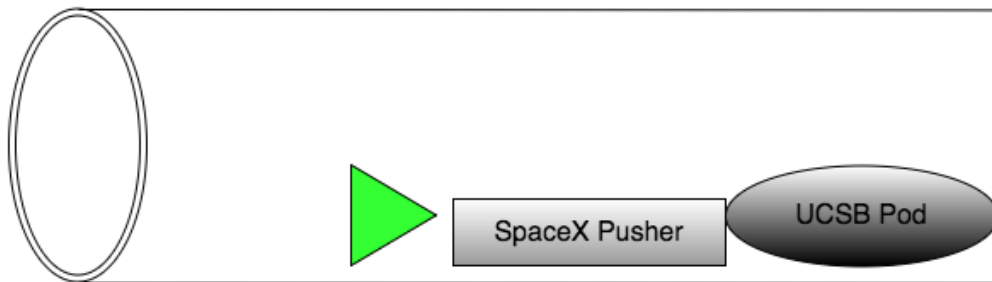
Phase lasts from pod entering tube to beginning of acceleration

Pre-Run Setup & Checks

- Establish web app connection
- Begin data monitoring / logging
- Calibrate sensors
- Run subsystem checks
- Engage maglev engines

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ACCELERATION PHASE



Phase lasts from pusher beginning movement until decoupling Pod

Navigation sensor data processed into unified measurement

- Photoelectric sensors (2) - highest accuracy
- Wheel tachometers (4) - highest precision
 - Combined tach value coerced to 100 ft. interval upon strip detection if within 15ft. window
- Accelerometers, pusher sensor provide additional checks

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COASTING PHASE



Phase lasts from pusher decoupling to start of braking

Pod must determine when to brake

- Timing profile required by SpaceX provides allowable window
 - Start - estimated time when coasting begins
 - End - last time to safely brake before end of tube
- Navigation determines distance within timing profile window
 - Braking begins at window end even if distance not reached

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BRAKING PHASE



Phase lasts from when braking starts until pod is stopped

- Distance Needed to Stop Pod
 $= \frac{1}{2} (\text{Pod Mass}) \times (\text{Pod Velocity})^2 / (\text{Braking Force})$

Braking Force Application

- Ideal actuation - dynamic adjustment of braking actuator force
- Timing profile end or emergency signal trigger full force braking

SERVICE PROPULSION PHASE



Phase lasts from pod stopping until end of tube is reached

- Maglev engines disabled and payload raised to lower drag
- Service propulsion assembly actuator lowered
- Service motor propels pod slowly to end of tube

Test run is then complete!

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EMERGENCY SITUATIONS

Conditions monitored

- Over-temperature in systems or power supply
- Over-current condition in systems
- Abnormal voltage conditions (cell imbalance, etc.)
- Imbalance in maglev motor speeds
- Imbalance in braking actuator force/position

Mitigation actions performed

- Affected system disabled
- Battery isolation by disconnecting relays

PROTOTYPE TESTING

- Four engine hovering [prototype](#)
- Was used for testing and code development
- Systems/phenomenon tested:
 - Vibrational environment
 - Subtrack and electronics heating
 - Telemetry sensors
 - Tachometers

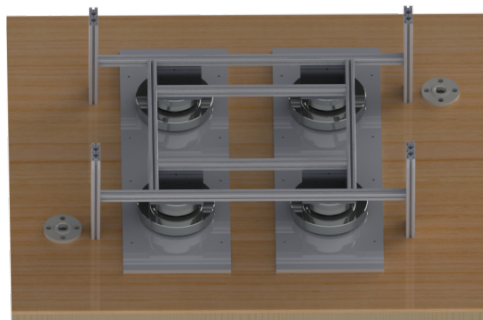


Figure: CAD Rendering of 4-Motor Prototype

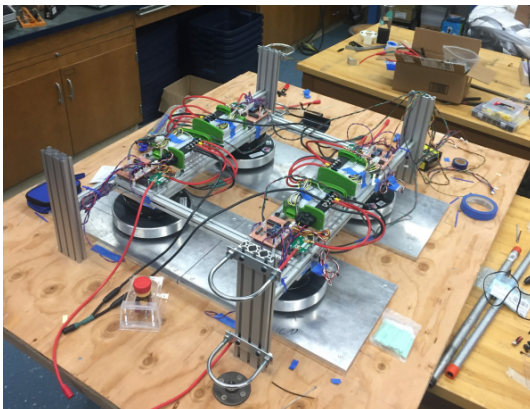


Figure: Working Prototype

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VACUUM TESTING

- SpaceX Tube reaches 0.15 psi
- 20" inner diameter bell jar vacuum chamber
- Low pressure performance tests:
 - Maglev engines
 - Motor controllers
 - LIPO batteries
 - PCB & peripheral boards
 - Sensors
 - Actuators



Figure: Bell vacuum chamber for testing

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CONCLUSION

SpaceX Testing Week and Hyperloop Competition II scheduled for Late August 2017

- Systems integration testing and preparation ongoing
- Our interns will continue system testing up to the competition date
- Opportunity for full scale on-site tests a week before the competition

ADVISORS

Thank you to Professor Johnson and Celeste for all of your help and guidance!

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NBK
The Motion Control Components